ATAR course examination, 2018

## Question/Answer booklet

## AVIATION

Please place your student identification label in this box

Student number: In figures


In words

## Time allowed for this paper

Reading time before commencing work:
Working time:
ten minutes
two and a half hours

## Materials required/recommended for this paper

To be provided by the supervisor
This Question/Answer booklet
Multiple-choice answer sheet

Number of additional answer booklets used (if applicable):

## To be provided by the candidate

Standard items: pens (blue/black preferred), pencils (including coloured), sharpener, correction fluid/tape, eraser, ruler, highlighters
Special items: non-programmable calculators approved for use in this examination, navigation plotter (or ruler and protractor), flight computer

## Important note to candidates

No other items may be taken into the examination room. It is your responsibility to ensure that you do not have any unauthorised material. If you have any unauthorised material with you, hand it to the supervisor before reading any further.

## Structure of the examination

The Aviation ATAR course examination consists of a written component and a practical (performance) component.

## Structure of this paper

| Section | Number of <br> questions <br> available | Number of <br> questions to <br> be answered | Suggested <br> working time <br> (minutes) | Marks <br> available | Percentage of <br> written <br> examination |
| :--- | :---: | :---: | :---: | :---: | :---: |
| Section One <br> Multiple-choice | 20 | 20 | 30 | 20 | 20 |
| Section Two <br> Short answer | 26 | 26 | 120 | 123 | 80 |

## Instructions to candidates

1. The rules for the conduct of the Western Australian external examinations are detailed in the Year 12 Information Handbook 2018. Sitting this examination implies that you agree to abide by these rules.
2. Answer the questions according to the following instructions.

Section One: Answer all questions on the separate Multiple-choice answer sheet provided. For each question, shade the box to indicate your answer. Use only a blue or black pen to shade the boxes. Do not use erasable or gel pens. If you make a mistake, place a cross through that square, then shade your new answer. Do not erase or use correction fluid/tape. Marks will not be deducted for incorrect answers. No marks will be given if more than one answer is completed for any question.

Section Two: Write your answers in this Question/Answer booklet.
3. Working or reasoning should be shown clearly when calculating or estimating answers.
4. You must be careful to confine your answers to the specific questions asked and to follow any instructions that are specific to a particular question.
5. Supplementary pages for planning/continuing your answers to questions are provided at the end of this Question/Answer booklet. If you use these pages to continue an answer, indicate at the original answer where the answer is continued, i.e. give the page number.

## Section One: Multiple-choice

This section has 20 questions. Answer all questions on the separate Multiple-choice answer sheet provided. For each question shade the box to indicate your answer. Use only a blue or black pen to shade the boxes. Do not use erasable or gel pens. If you make a mistake, place a cross through that square, then shade your new answer. Do not erase or use correction fluid/tape. Marks will not be deducted for incorrect answers. No marks will be given if more than one answer is completed for any question.

Suggested working time: 30 minutes.

1. Which of the following distribution of forces applies to an aircraft in a steady state climb?
(a) L $>$ W and T $<$ D
(b) $\mathrm{L}=\mathrm{W}$ and $\mathrm{T}=\mathrm{D}$
(c) L $>$ D and $W<T$
(d) L $<$ W and T $>$ D
2. A high-speed stall is most likely to occur when
(a) entering a high-speed descent.
(b) entering a high-speed climb.
(c) in a high-speed level turn at an angle of bank greater than $20^{\circ}$.
(d) quickly levelling out after a high-speed steep descent.
3. Wing loading of an aircraft is calculated using
(a) wing area and weight.
(b) wing area and the calculated differential pressure between the top and bottom of the wing.
(c) lift divided by weight.
(d) the difference between the manufacturer's maximum take-off weight and the aircraft's empty weight at the time of manufacture.
4. Which of the following would qualify as high-lift devices when fitted to an aircraft?
(a) spoilers
(b) low aspect ratio wings
(c) elevons
(d) vortex generators
5. In addition to the last light tables, which of the following are required to calculate first light as a local mean time at a given place on the surface of the Earth?
(a) longitude, month and year
(b) latitude, day and month
(c) day, month and year
(d) longitude and the season
6. Which of the following is required for the formation of a convection thunderstorm to occur?
(a) orographic uplift
(b) a strong surface heating
(c) the formation of a radiation fog
(d) the convergence of two similar air masses
7. Vfe is the
(a) maximum permissible speed for flight with flaps extended.
(b) maximum speed for selecting full flap.
(c) minimum speed required for operating the flaps.
(d) maximum speed for flying for best endurance.
8. A piston-engined aircraft wanting to achieve best endurance needs to fly
(a) as high as possible at the speed for maximum surplus thrust.
(b) as low as practical with the speed for minimum thrust.
(c) as low as practical with the speed for minimum power.
(d) as high as practical while maintaining the lowest possible speed for minimum straight and level flight.
9. When supersonic airflow changes direction while flowing over the upper surface of a wing, it will
(a) decrease in temperature.
(b) increase in density.
(c) increase in pressure.
(d) decrease in speed.
10. Extent of cloud is reported in aviation using the term OKTAS. This refers to
(a) whether the cloud is okay to fly through on a scale of 1 to 8 and the speed (TAS) at which it is moving.
(b) the cloud coverage of the visible sky on a scale of 1 to 8 .
(c) the direction of cloud from the observer using $015^{\circ}$ intervals, for example 1 OKTAS is cloud between $360^{\circ} \mathrm{M}$ and $015^{\circ} \mathrm{M}$.
(d) the type of cloud and likelihood of severe windshear and turbulence.
11. The greatest hazard that thunderstorms in the vicinity of an airport pose to landing aircraft is the possible presence of
(a) heavy rain.
(b) hail and snow.
(c) microbursts.
(d) severe turbulence.
12. The position of the North Magnetic pole is
(a) continually varying but the annual change will not normally cause a significant navigational error and is ignored in Australia.
(b) continually varying and this variation must always be adjusted for whenever it is applied to a true track.
(c) a fixed point on the earth's surface.
(d) continually varying and deviation must be accounted for whenever it is applied to a true track.
13. A force applied to a gyroscopic instrument that acts $90^{\circ}$ in the direction of rotation is known as
(a) wander.
(b) tilt.
(c) drift.
(d) precession.
14. Hypermetropia is the inability of a person to
(a) maintain a normal body temperature.
(b) focus their vision on close objects.
(c) hear sounds from a distance.
(d) maintain a normal blood pressure.
15. If an aircraft is tracking $045^{\circ} \mathrm{T}$ and the variation is $6^{\circ} \mathrm{E}$, the magnetic track is
(a) $051^{\circ} \mathrm{M}$.
(b) $045^{\circ} \mathrm{M}$.
(c) $039^{\circ} \mathrm{M}$.
(d) cannot be calculated without the wind direction and strength.
16. The tail rotor on a helicopter is used to
(a) nullify the Coriolis effect.
(b) make the helicopter fly sideways.
(c) eliminate gyroscopic precession.
(d) counteract any torque effect.
17. Updrafts are a feature of thunderstorms. In the 'life' of a thunderstorm, updrafts are most likely to exist in
(a) the developing and mature stages.
(b) the mature and dissipating stages.
(c) the developing stage only.
(d) all stages.
18. Which of the following statements is true regarding an aircraft that is on a collision course with your aircraft? The other aircraft will appear to
(a) move downward in the windscreen relative to your aircraft.
(b) move upward in the windscreen relative to your aircraft.
(c) remain stationary in the windscreen relative to your aircraft.
(d) move in any direction relative to your aircraft.
19. Radiation fog in the hills will normally clear and rise up to become
(a) stratus cloud.
(b) nimbostratus cloud.
(c) cumulus cloud.
(d) cumulonimbus cloud.
20. The main issue of an ageing aircraft fleet in Australia is related to the impact on safety resulting from which of these concerns?
(a) new pilots trained on more modern aircraft may not being able to manage instruments/operating systems of older aircraft
(b) older aircraft are not compatible with the new fuels being introduced
(c) increased probability of aircraft accidents as a result of structural failure due to metal fatigue
(d) the lower costs associated with buying and insuring newer aircraft when compared with older aircraft and the losses aircraft owners may face if they cannot sell their old aircraft

## End of Section One

## Section Two: Short answer

This section has 26 questions. Answer all questions. Write your answers in the spaces provided.
Supplementary pages for planning/continuing your answers to questions are provided at the end of this Question/Answer booklet. If you use these pages to continue an answer, indicate at the original answer where the answer is continued, i.e. give the page number.

Suggested working time: 120 minutes.

## Question 21

Given the following data:

- airfield elevation 1525 ft
- QNH 1007
- OAT $33^{\circ} \mathrm{C}$.

Calculate airfield pressure altitude. Show all workings.
$\qquad$
$\qquad$
$\qquad$
$\qquad$

## Question 22

Given the following data:

- airfield pressure height 175 ft
- QNH 1001
- OAT $35^{\circ} \mathrm{C}$.

Calculate airfield density altitude. Show all workings.
$\qquad$
$\qquad$
$\qquad$
$\qquad$

## Question 23

A pilot is about to order fuel for her aircraft. The following data apply:

- fuel tank capacity 230 L
- AVGAS specific gravity 0.72 .

In order for the aircraft to remain within performance limitations, the maximum additional fuel that can be loaded into the tank cannot exceed 93 kg .

Determine the maximum number of whole litres that can be added so as not to exceed these limitations. Show all workings.
$\qquad$
$\qquad$
$\qquad$
$\qquad$

## Question 24

An aircraft is tracking $175^{\circ} \mathrm{M}$ and maintaining 120 kt TAS.
Actual winds $125^{\circ} \mathrm{M} / 25 \mathrm{kt}$.

Assume no compass errors exist.

Use your flight computer to resolve the following:
(a) What heading would be required to be flown to maintain the planned track?
$\qquad$
(b) What ground speed would the aircraft be maintaining?
$\qquad$

## Question 25

An aircraft is flying between two known points with the following data:

- distance 180 nm
- TAS 120 kt
- forecast winds $235^{\circ} \mathrm{M} / 20 \mathrm{kt}$
- flight planned track $090^{\circ} \mathrm{M}$
- heading flown $095^{\circ} \mathrm{M}$.

After travelling 75 nm , the aircraft has drifted 5 nm to the right of track.
(a) Using only the 1 in 60 rule, determine the actual Track Made Good (TMG). Show all workings.
$\qquad$
$\qquad$
$\qquad$
$\qquad$
(b) On the basis of the data provided, describe the most plausible reason why the aircraft is off track at the 75 nm point.
$\qquad$
$\qquad$

Answer the question parts below using the PA-32RT performance chart on page 11 and the following data:

- outside air temperature $0^{\circ} \mathrm{C}$
- pressure altitude 4000 ft
- power setting $55 \%$.
(a) Determine the TAS for the flight conducted for best economy. Show all workings on the chart.
(b) Determine the new TAS if the aircraft was set up to fly $75 \%$ power for best power while still maintaining the advised altitude.
$\qquad$
(c) The RPM and Manifold Absolute Pressure (MAP) limitations for Maximum Continuous Power (M.C.P) at 8000 ft pressure altitude in ISA would be:

RPM: $\qquad$
MAP:
The resulting TAS would be closest to: $\qquad$

A VFR aircraft is travelling south holding a constant heading of $180^{\circ} \mathrm{M}$. The aircraft passes over its initial waypoint at UTC 0425 and has estimated its next waypoint at UTC 0615. The winds are forecast to remain constant and the distance between waypoints is 260 nm .
(a) Calculate the planned ground speed. Give the correct unit of measurement.
$\qquad$
$\qquad$
(b) The aircraft arrives over its second waypoint at UTC 0600. Calculate the actual ground speed (to the nearest whole unit) achieved for this flight.
$\qquad$
$\qquad$
$\qquad$
$\qquad$

Using the End of daylight chart below, calculate last light in LMT on 10 October for Albany ( $34^{\circ} 57^{\prime} \mathrm{S}, 117^{\circ} 49^{\prime} \mathrm{E}$ ). Show all workings.
$\qquad$
$\qquad$

End of daylight


## Question 29

The following data apply to an aircraft:

- TAS 120 kt
- altitude 6500 ft
- fuel policy - fixed reserve 45 minutes
- taxi fuel 5 litres
- climb and descent requirements are to be ignored
- planned fuel consumption for this flight 45 litres per hour.

Determine the minimum onboard fuel requirements (in litres) at start up for a flight 276 nm at a planned ground speed of 150 kt . Show all workings.
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$

## Question 30

The following data apply to an aircraft:

- heading $180^{\circ} \mathrm{M}$ magnetic variation nil
- navigation instrumentation DME - 35 nm
- VOR information

OBI 180
flag shows FROM
CDI - centred.
Plot the aircraft's position and orientation as accurately as possible using your navigational plotter and WAC scale ruler (1: 1000000 ) on the basic orientation diagram below.

Note: Circle represents collocated VOR/DME.


Use your flight computer to calculate and complete the following table.

| TAS | Track <br> magnetic | Winds <br> magnetic | Angle of <br> drift | Direction <br> of drift | Heading | GS |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 145 kt | $140^{\circ}$ | $025 / 25$ |  |  |  |  |

## Question 32

Using the information table below and the Take-off and Performance charts provided on pages 17-19, complete the following question parts that relate to a proposed flight in a Piper PA-32RT-300T Turbo Lance from Alpha to Bravo.

| Conditions | Alpha | Bravo | En route |  |
| :--- | :--- | :--- | :--- | :--- |
| Pressure height | Sea level | 2000 ft | Distance | 355 nm |
| Runway surface | Long dry grass | Short wet grass | Cruise level | Alt 4500 ft |
| Runway slope | $2^{\circ}$ down | nil | Fuel flow | 13.7 gph |
| Take-off weight | 1630 kg |  | Headwind | Nil |
| Temperature | $10^{\circ} \mathrm{C}$ | $15^{\circ} \mathrm{C}$ | OAT | $10^{\circ} \mathrm{C}$ |
| Wind | nil | nil | TAS | 145 kt |

(a) Determine the minimum take-off distance required at Alpha. Show all workings clearly on the appropriate chart.
$\qquad$
(b) Determine climb, cruise and descent data to complete the table below to find the

- total flight time.
- total flight fuel required, i.e. excluding reserves, taxi and unusable fuel.

Ignore all winds in the climb, cruise and descent. Show all workings clearly on the appropriate chart.

|  | Climb | Cruise | Descent | Total |
| :--- | :--- | :--- | :--- | :---: |
| Fuel (gal) |  |  |  |  |
| Time (min) |  |  |  |  |
| Distance (nm) |  |  |  | 355 nm |

Question 32 (continued)



Weight and balance are important parameters for aircraft.
Complete the table below for a Piper PA-32RT-300T Turbo Lance aircraft to show the weight, position of the centre of gravity and moment at zero fuel weight.

| Position | Weight (lb) | Arm (in) | Moment (Ib/in) |
| :--- | ---: | ---: | ---: |
| Aircraft (BEW) | 2335.8 |  | 195086.0 |
| Front row | 322.0 | 85.5 |  |
| Centre row | 225.0 | 118.1 |  |
| Rear row | 140.0 | 157.6 |  |
| Forward <br> baggage | 5.0 | 42.0 |  |
| Aft <br> baggage | 50.0 | 178.7 |  |
| Zero fuel weight |  |  |  |

An aircraft needs to be balanced in all stages of flight.
(a) Complete the table below to show the position of centre of gravity and moment for the aircraft at ramp prior to taxi.

| Position | Weight (Ib) | Arm (in) | Moment (Ib/in) |
| :--- | ---: | ---: | ---: |
| Zero fuel weight | 2950 | 93.0 | 274350 |
| Fuel | 325 | 93.6 | 30420 |
| Ramp weight | 3275 |  |  |

(b) In which direction does the loading of fuel move the centre of gravity on this occasion?

Justify your answer.
Direction of movement: $\qquad$
Justification: $\qquad$
$\qquad$
$\qquad$
$\qquad$

Given a time of 1129 Local Mean Time (LMT) and a position of $15^{\circ} 25^{\prime} \mathrm{S} 134^{\circ} 20^{\prime}$ E, use the Conversion of Arc to Time chart below to convert LMT to Coordinated Universal Time (UTC). Show all workings.

| CONVERSION OF ARC TO TIME |  |  |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| DEGREES |  |  |  |  |  | MINUTES |  |  |  |  |  |
| Long <br> Deg | Time |  | $\begin{aligned} & \text { Long } \\ & \text { Deg } \end{aligned}$ | Time |  | Long Min | Time |  | Long <br> Min | Time |  |
|  |  |  |  | Hour Min |  |  | Min | Sec |  | Min | Sec |
| 110 | 7 | 20 | 140 | 9 | 20 | 0 | 0 | 00 | 30 | 2 | 00 |
| 111 | 7 | 24 | 141 | 9 | 24 | 1 | 0 | 04 | 31 | 2 | 04 |
| 112 | 7 | 28 | 142 | 9 | 28 | 2 | 0 | 08 | 32 | 2 | 08 |
| 113 | 7 | 32 | 143 | 9 | 32 | 3 | 0 | 12 | 33 | 2 | 12 |
| 114 | 7 | 36 | 144 | 9 | 36 | 4 | 0 | 16 | 34 | 2 | 16 |
| 115 | 7 | 40 | 145 | 9 | 40 | 5 | 0 | 20 | 35 | 2 | 20 |
| 116 | 7 | 44 | 146 | 9 | 44 | 6 | 0 | 24 | 36 | 2 | 24 |
| 117 | 7 | 48 | 147 | 9 | 48 | 7 | 0 | 28 | 37 | 2 | 28 |
| 118 | 7 | 52 | 148 | 9 | 52 | 8 | 0 | 32 | 38 | 2 | 32 |
| 119 | 7 | 56 | 149 | 9 | 56 | 9 | 0 | 36 | 39 | 2 | 36 |
| 120 | 8 | 00 | 150 | 10 | 00 | 10 | 0 | 40 | 40 | 2 | 40 |
| 121 | 8 | 04 | 151 | 10 | 04 | 11 | 0 | 44 | 41 | 2 | 44 |
| 122 | 8 | 08 | 152 | 10 | 08 | 12 | 0 | 48 | 42 | 2 | 48 |
| 123 | 8 | 12 | 153 | 10 | 12 | 13 | 0 | 52 | 43 | 2 | 52 |
| 124 | 8 | 16 | 154 | 10 | 16 | 14 | 0 | 56 | 44 | 2 | 56 |
| 125 | 8 | 20 | 155 | 10 | 20 | 15 | 1 | 00 | 45 | 3 | 00 |
| 126 | 8 | 24 | 156 | 10 | 24 | 16 | 1 | 04 | 46 | 3 | 04 |
| 127 | 8 | 28 | 157 | 10 | 28 | 17 | 1 | 08 | 47 | 3 | 08 |
| 128 | 8 | 32 | 158 | 10 | 32 | 18 | 1 | 12 | 48 | 3 | 12 |
| 129 | 8 | 36 | 159 | 10 | 36 | 19 | 1 | 16 | 49 | 3 | 16 |
| 130 | 8 | 40 |  |  |  | 20 | 1 | 20 | 50 | 3 | 20 |
| 131 | 8 | 44 |  |  |  | 21 | 1 | 24 | 51 | 3 | 24 |
| 132 | 8 | 48 |  |  |  | 22 | 1 | 28 | 52 | 3 | 28 |
| 133 | 8 | 52 |  |  |  | 23 | 1 | 32 | 53 | 3 | 32 |
| 134 | 8 | 56 |  |  |  | 24 | 1 | 36 | 54 | 3 | 36 |
| 135 | 9 | 00 |  |  |  | 25 | 1 | 40 | 55 | 3 | 40 |
| 136 | 9 | 04 |  |  |  | 26 | 1 | 44 | 56 | 3 | 44 |
| 137 | 9 | 08 |  |  |  | 27 | 1 | 48 | 57 | 3 | 48 |
| 138 | 9 | 12 |  |  |  | 28 | 1 | 52 | 58 | 3 | 52 |
| 139 | 9 | 16 |  |  |  | 29 | 1 | 56 | 59 | 3 | 56 |

An aircraft is loaded and balanced for flight at zero fuel weight and for take-off weight as shown in the table below.

|  | Weight (lb) | Arm (in) | Moment (Ib/in) |
| :--- | ---: | ---: | ---: |
| Gross <br> Zero fuel weight | 2950 | 92.8 | 273760 |
| Gross <br> Take-off weight | 3300 | 94.8 | 312840 |

A passenger has arrived with unexpected extra baggage that can only go in the aft baggage compartment.

Using the formula provided, determine the maximum theoretical amount of additional baggage that could be added to the aft baggage compartment for both zero fuel weight and take-off weight to load the aircraft to a desired maximum aft centre of gravity position of 96 in. Show all workings.

Weight to add $=\quad$ Gross weight $x$ desired change of centre of gravity Distance between loading station and desired centre of gravity

The aft baggage loading station is located at 178.7 in.
The aft limit for this aircraft is not to exceed 96 in.
Ignore all maximum weight limitations for this question.
Zero fuel weight: $\qquad$
$\qquad$
$\qquad$
$\qquad$

Take-off weight: $\qquad$
$\qquad$
$\qquad$
$\qquad$
(a) Use the centre of gravity vs weight chart provided to plot and label both the zero fuel weight (1) and proposed zero fuel weight (2).

|  | Weight (Ib) | Arm (in) | Moment (Ib/in) |
| :--- | ---: | ---: | ---: |
| Zero fuel weight (1) | 3100 | 95 | 294500 |
| Proposed zero fuel <br> weight (2) | 3400 | 97 | 329800 |

Centre of Gravity vs Weight Envelope

(b) Purely on the basis of your observation from the positions plotted on the graph, determine the maximum permissible zero fuel weight for this aircraft if the weight was adjusted so the aircraft remained loaded to the maximum aft limit only.

## Question 38

Anhedral is an aerodynamic design feature fitted to some aircraft.
(a) State the purpose of this design feature.
$\qquad$
$\qquad$
(b) Explain why an anhedral would be required to be fitted to a high wing heavy transport aircraft with sweptback wings such as the Antonov AN-225. Use diagrams to assist with your answer.
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$

Calibrated air speed (CAS) is a term referred to in the aviation environment.
(a) What is CAS?
$\qquad$
$\qquad$
(b) What two factors need to be taken into consideration in order to calculate a CAS on a light general aviation fixed-wing propeller driven aircraft?

One: $\qquad$
$\qquad$
Two: $\qquad$

## Question 40

On the basis of the synoptic chart below, identify the features and extract the relevant information to answer the question parts on page 27.

(a) What is the pressure at position $A$ ?
$\qquad$
(b) Which meteorological feature is shown by the dashed line between positions E and F ?
(c) The frontal system passing over central New Zealand, below position F, is moving in which direction and at what speed?

Direction: $\qquad$
Speed: $\qquad$
(d) Describe the frontal system incorporating the line from position C to position D. (2 marks)
$\qquad$
$\qquad$
(e) Identify the system and describe conditions associated with the feature between positions $B$ and $C$ as it passes over the south-west of Western Australia.
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$

Good ergonomic design performs a pivotal role in assisting with aviation safety.
(a) What is meant by the term 'ergonomics'.
$\qquad$
$\qquad$
(b) Using an example, explain how the implementation of ergonomic principles can enhance safety.
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$

## Question 42

Cyclones are a serious hazard to aviation in the Southern Hemisphere.
(a) Outline three prerequisite conditions necessary for the formation of a cyclone in the Southern Hemisphere.

One: $\qquad$
$\qquad$
Two: $\qquad$
$\qquad$
Three: $\qquad$
$\qquad$
(b) Describe conditions within a cyclone and the area immediately around it (out to 50 nm ) as the cyclone travels over an ocean and overland. In each of your answers outline what happens to its intensity and why these changes occur.
(i) Ocean:
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
(ii) Overland:
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$

Visual illusions and other natural tendencies of the human eye can cause problems for pilots in flight. Explain the effects of the following, including how they occur and how they may be reasonably managed.
(a) Empty-field myopia:
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
(b) Auto kinesis:
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
(c) False horizon:
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$

Some aircraft are fitted with an engine manifold pressure gauge.
(a) What does the manifold pressure gauge actually measure and where is the measurement taken?
$\qquad$
$\qquad$
$\qquad$
$\qquad$
(b) Describe how manifold air pressure would be adjusted/controlled by the pilot. (2 marks)
$\qquad$
$\qquad$
$\qquad$
$\qquad$

Carbon monoxide poisoning poses a real risk to pilots.
(a) Explain how carbon monoxide affects the body.
$\qquad$
$\qquad$
$\qquad$
$\qquad$
(b) Give two symptoms a pilot could experience if exposed for prolonged periods to carbon monoxide.

One: $\qquad$
$\qquad$
Two: $\qquad$
$\qquad$
(c) How can a pilot detect carbon monoxide in flight?
$\qquad$
$\qquad$
(d) What would be the best immediate treatment for a person suffering from carbon monoxide poisoning?
$\qquad$

## Question 46

Research continues into practical alternative means of operating an aircraft's motor.
Discuss the factors influencing the ongoing development of both alternative fuels and implementation of electric aircraft motors. Highlight the likely positive and negative impact these developments might have on the aviation industry.
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$
$\qquad$

End of questions

Supplementary page
Question number:

## Supplementary page

Question number:

## ACKNOWLEDGEMENTS

## Question 26

Question 28

Question 32 PA-32RT take-off weight chart from: Yeo, M., Bowers, G., \& Bennett, K. (2001). Handbook of flight (2nd ed.). Perth: WestOne Services, p. 149. Not for operational purposes.
PA-32RT fuel, time and distance to climb chart from: Yeo, M., Bowers, G., \& Bennett, K. (2001). Handbook of flight (2nd ed.). Perth: WestOne Services, p. 169. Not for operational purposes.
PA-32RT fuel, time and distance to descend chart from: Yeo, M., Bowers, G., \& Bennett, K. (2001). Handbook of flight (2nd ed.). Perth: WestOne Services, p. 171. Not for operational purposes

Question 35 Conversion of ARC to time chart from: Airservices Australia. (2016). Aeronautical information package (AIP): AIP book (General: GEN 2.7 - 7: 25 Nov 04: Conversion of ARC to time). Canberra: Airservices Australia. Retrieved May, 2016 from www.airservicesaustralia.com/aip/ aip.asp?pg=20\&vdate=26-May-2016\&ver=2
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## Question 40

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